

Numerical Modelling of Compressible Flows Using Pressure-Correction Algorithms

Frédéric Archambeau, Jean-Marc Hérard, Jérôme Laviéville

EDF-R&D, MFEE, 6 quai Watier, 78400, Chatou, France (frederic.archambeau@edf.fr, jean-marc.herard@edf.fr, jerome-marcel.lavieville@edf.fr)

Compressible flows with shock solutions are encountered in a wide variety of situations, from aeronautics to power generation industry. It is well known that Godunov-type methods provide very good approximations for such flows ([6], [7] for example). However, some existing industrial software that are based on pressure correction methods may have to deal with such configurations. Two pressure correction schemes are considered ([1] and [10], [8]). We show numerically on several academic test-cases that they can capture the exact shock solutions of Euler equations provided conservativity in time and space is ensured (see also [5] and [9]). We finally compare the pressure correction schemes to the approximate Godunov scheme VFRoe-ncv [2] on a test-case with a strong local and constant heat source term. This test-case, for which no analytical solution is available, represents the Joule effect due to an unwanted electric arc in an electrical power transformer (see [3]). If conservativity is ensured, all schemes converge towards the same solution as the mesh is refined.

System of equations

Ø

We consider the Euler equations with perfect gas thermodynamics under form (1) or (2):

- 1	$\partial_t \rho + \partial_x (\rho u)$	=	0	- 1	$\partial_t \rho + \partial_x (\rho u)$	=	0
(1)	$\partial_t(\rho u) + \partial_x(\rho u u) + \partial_x P$	=	0	(2)	$\partial_t(\rho u) + \partial_x(\rho u u) + \partial_x P$	_	0
	$\partial_t(\rho H) + \partial_x(\rho u H)$	=	$\partial_t P + \Phi$		$\partial_t(\rho E) + \partial_x(\rho u E + u P)$	=	Φ

An iterative pressure-correction algorithm: NLK

At each time step, the algorithm solves system (1) in three steps with sub-iterations:



Update mass flow rate for next m iteration $Q_{m+1}^{n+1} = q_{m,k_{max}}^{n+1}$

Conservativity in time is ensured at convergence of the m- and k-cycles since:

$$\frac{u^{*}_{m_{max}} - u^{n}}{\Delta t} - u^{*}_{m_{max}} \partial_{x} Q^{n+1}_{m_{max}} = \frac{\rho^{n+1} u^{n+1} - \rho^{n} u^{n}}{\Delta t}$$

A non iterative pressure-correction algorithm: SLK

At each time step, the algorithm solves system (2) in three steps without sub-iteration:

$$\begin{array}{lll} \textbf{Density step} & \frac{\rho^{n+1}-\rho^n}{\Delta t} + \partial_x \left(q^n - \Delta t \,\theta \,\beta \,\partial_x \,s^n\right) - \partial_x \left(\Delta t (c^2 \,\partial_x \,\rho^{n+1})\right) = 0 \\ & Update the convective mass flux & q_{ac}^{n+1} = q^n - \Delta t \, \left(c^2 \,\partial_x \,\rho^{n+1} + \theta \,\beta \,\partial_x \,s^n\right) \\ \textbf{Velocity step} & \rho^n \, \frac{u^{n+1}-u^n}{\Delta t} - u^{n+1} \,\partial_x q_{ac}^{n+1} + \partial_x \left(q_{ac}^{n+1} \,u^{n+1}\right) = -\partial_x P^n \\ \textbf{Energy step} & \rho^n \, \frac{E^{n+1}-E^n}{\Delta t} - E^{n+1} \,\partial_x q_{ac}^{n+1} + \partial_x \left(q_{ac}^{n+1} \,\left(E^{n+1} + \frac{P^n}{\rho^{n+1}}\right)\right) = \Phi \\ & Update the pressure \end{array}$$

Conservativity in time is ensured thanks to the definition of the convective mass flux.

Analytical test-cases without heat source term

We consider a contact discontinuity, the Sod shock tube, a double expansion, a double shock and a sonic shock tube. The behaviour of so-called first order schemes is recovered ([4]): NLK operated with a sufficient number of subcycles and SLK both capture the exact solution computed following [11]. We outline that NLK operated with too few subcycles is no more conservative in time and does not converge towards the exact solution for cases involving a shock.



<u>Fig. 1</u>: evolution of the L⁻-error for the density on the analytical test-cases without heat source term. Plots show the behaviour for SLK (left) and for NLK with k_{max} =20 and m_{max} =1 (right).



SLK algorithm (160 cells)
 SLK algorithm (1,280 cells)
 SLK algorithm (1,240 cells)
 SLK algorithm (10,240 cells)
 Analytical solution

<u>Fig 2</u>: density profile for the sonic shock tube [12] calculated with SLK on different meshes. No spurious phenomenon is observed at the location of the initial discontinuity.

Test-case with a constant heat source term





<u>Fig 3</u>: evolution of the L^1 -difference to VFRoe-ncv for the density (test-case with a constant heat source term).

<u>Fig 4</u>: illustration of the profiles of the density, velocity, pressure and energy obtained for the test-case with a constant heat source term.

[1] F. Archambeau, N. Méchitoua, M. Sakiz, Code_Saturne: A finite volume code for the computation of turbulent incompressible flows - Industrial applications, International Journal on Finite Volumes, http://averoes.math.univ-paris13.fr/, Volume 1, 2004.

[2] T. Buffard, T. Gallouët, J.-M. Hérard, A sequel to a rough Godunov scheme: application to real gases, Computers and Fluids, vol. 29, n. 7, pp. 813-847, 2000.

- [3] A. Douce, C. Delalondre, H. Biausser, J.B. Guillot, Numerical modelling of an anodic metal bath heated with an Argon transferred arc, *The Int. Iron and Steel Institute of Japan*, vol. 43 (8), pp. 1128-1135, 2003.
 [4] T. Gallouët, J.-M. Hérard, N. Seguin, Some recent finite volume schemes to compute Euler equations using real gas EOS, *Int. Journal for Numerical Methods in Fluids*, vol. 39, n. 12, pp. 1073-1138, 2002.
 [5] T. Gallouët, J.-M. Hérard, N. Seguin, A hybrid scheme to compute contact discontinuities in one dimensional Euler systems, *Math. Model. and Numerical Analysis*, vol. 36, pp. 1133-1159, 2002.
- [6] E. Godlewski, P.-A. Raviart, Numerical approximation of hyperbolic systems of conservation laws, New York, Springer-Verlag, 1996.

[8] A. Guelfi, D. Bestion, M. Boucker, P. Boudier, P. Fillion, M. Grandotto, J.-M. Hérard, E. Hervieu, P. Péturaud, NEPTUNE - A new software platform for advanced nuclear thermal hydraulics, Nuclear Science and

Engineering, vol. 156, pp. 281-324, 2007. [9] X. Hou, P.G. Le Floch, Why non conservative schemes converge to wrong solutions: error analysis, *Mathematics of Computation*, vol. 62, pp. 497-530, 1994.

- [10] P. Mathon, F. Archambeau, J.-M. Hérard, Implantation d'un algorithme compressible dans Code_Saturne, EDF internal report H183-03/016, in French, unpublished, 2003.
- [11] J. Smoller, Shock waves and reaction-diffusion equations, New York, Springer-Verlag, 1983.
- [12] E. Toro, Riemann solvers and numerical methods for fluid dynamics, Berlin, Springer-Verlag, 1999.

^[7] S.K. Godunov, Finite difference method for numerical computation of discontinuous solutions of the equations of fluid dynamics, Mat. Sb., vol. 47, pp. 271-300, 1959.